

# EGR270 – MATERIALS LABORATORY

Lab 3 - Strain Gages

Assigned date: Jan 29th

Due Date(s): Week of Feb 5<sup>th</sup>

## Description

Students explore the calibration of load cells and fundamentals of weight measurement using NIST certified weights. Students also apply this knowledge to the measurement of lift of a small quadcopter.

## **Learning Objectives**

- Practice a procedure of calibrating a measurement device (cantilever beam load cell).
- Practice a procedure of calibrating weights to be used as local standards.
- Use local standards to estimate the uncertainty (error) in a scale.
- Explain how to measure the thrust output of a quadcopter and possible difficulties.

## Grading

This assignment will count towards the worksheet portion of the course grade. Graded will be based on completeness and reasonableness of your write-up. See attached letter from the faculty to you for details.

## **Submissions and Late Policy**

Assigned work will be collected exclusively during the first 5 minutes of lab next week. No other mode of submission is acceptable unless expressly specified by the professor. No late submissions are accepted.

# **Overall (non-detailed) Procedures**

There are two primary "segments" to this week's lab. In Segment 1, you are acting as a cal/cert technician (someone who performs calibration to certify measurement items). In Segment 2, you are acting as a mechanical engineering testing the lift characteristics of a small quadcopter. The general procedure:

Segment 1 (calibration and certification):

- 1) Calibrate two cantilever beam load cells using calibrated and certified weights traceable to NIST.
- 2) Use the calibrated cantilever beam load cells to calibrate four different weights (these become "local standards" useable for calibrating other load cells, but with greater error and uncertainty than the certified weights used in step 1).

Segment 2 (test engineer, determine flight characteristic of a quadcopter):

- 3) Use the newly certified local standards to estimate uncertainty (error) in two different quadcopter test pads (a digital scale and a polymer cantilever beam).
- 4) Measure lift produced by a quadcopter using both quadcopter test stands.



January 29, 2018

Students Materials Science Laboratory 5000 N. Willamette Blvd. Portland, OR 97203

Dear Students,

We request your assistance in two measurement tasks. First, we need you to calibrate a cantilever beam load cell and weights. We would like to know the calibration factor (the relationship between strain and applied weight). Then we request your assistance in evaluating the performance of a small quadcopter. We would like to know what the maximum load that this quadcopter can lift. Please test the quadcopter provided in the lab. We have provided you with a set of NIST certified weights, a set of uncertified weights (that you will calibrate), as well as two quadcopter test stands.

It may interest you that a fellow engineer thinks that a simple scale will not be sufficient to measure the thrust of the quadcopter. He has already done some work in this area, and I have attached his notes. Please investigate this claim while you carry out this performance evaluation and let me know if you agree with his conclusion. Make sure you back up your claims with solid evidence, such as percent-differences between any measurement methods you may use.

For lab class next week, please submit 1-2 paragraphs (along with any supporting figures and tables) making an engineering recommendation as to which measurement system you recommend. This is not a technical letter or memo. Please clearly indicate your name, date, class, and lab section on your submission. Make sure to consult the sections "*Things I need to investigate*" and "*Comments from conversation with the boss*" for items that you should be including your submission. Please note, since a significant part of this laboratory is calibration of a load cell, you must include a calibration curve for both the aluminum and steel cantilever beams. Be sure to include a linear regression line that shows both the R<sup>2</sup> value (an indication of how well the line fits the data point, R<sup>2</sup>=1.0 means the data lies directly on the line) and the equation for the line (the slope is the calibration factor). Like all figures graphs, discuss the importance of the graph.

Thank you for help in this matter. If you have any questions, do not hesitate to contact one of your lab professors (see syllabus). Please note, I've provided a blank data sheet on the reverse side of this letter.

Sincerely,

(electronic)

Your Professors

Enclosed: one (1) set of NIST certified weights, four (4) uncertified weights, two (2) uncalibrated straingaged cantilevers, one (1) quadcopter, one (1) uncalibrated digital scale, one (1) instrumented helipad, clean gloves for each student, one (1) background document discussing calibration and quadcopter measurement.



# **CALIBRATION NOTES**

#### Definitions:

**Load cell:** A load cell is a specific type of transducer that is used to create an electrical signal that is directly proportional to the force being measured.

**Transducer**: A transducer is a measuring device that converts variations in a physical quantity into a measurable quantity. A mercury-in-glass thermometer is a simple example of a transducer, and so is a strain gage.

**Measurement Uncertainty**: All engineering data is based on measurements; yet no measurement is perfect. Even the most accurate and precise measurement is an *estimation* of the true value. Measurement uncertainty is a statistically derived estimate for "error". "Error" is never known exactly, it is an estimate.

**True value**: if you could measure something perfectly, you would be measuring its true value. All measurements are approximations to the true value (the error is the difference between the measured value and true value).

**Calibration**: Calibration is the process of comparing values indicated by a measuring device with those of a calibration standard of known accuracy. Calibration allows for a certain degree of confidence in the measuring device's ability. This lab is meat to demonstrate the principles of calibration in a simplified way. Calibrations performed by calibration labs usually requires very precise procedures in a controlled environment.

**Certification of measuring devices**: Certification is the documentation showing the results of calibration by qualified personnel.

Cal Cert: The abbreviated version of "calibration and certification".

**Hierarchy of Standards**: **primary** standards are the definitions of the measurement. For example, there is only one primary standard for the definition of a kilogram (buried in a vault in France), and the speed of light (distance light travels in 1/299,792,458 seconds in a vacuum) is the primary standard for the meter. **Secondary standards** are produced by comparing them with the primary standard, and therefore, are very close approximations. Standards that are created by comparison to secondary standards are known as tertiary, **local**, or working standards.

#### Notes on Strain Gages:

There are several different types of strain gages used for measuring strain. For this experiment, you will be using etched foil strain gages. In an etched foil strain gage, a very thin piece of metal is etched and placed on a plastic film carrier. This is known as a bonded-foil strain gage. This assembly is then carefully glued onto the test specimen whose strain is to be measured. As the metal foil in the strain gage is stretched or compressed along with the test specimen, its electrical resistance changes. Depending on the metal used, very large strains of up to 50,000  $\mu\epsilon$  (50,000  $\times$  10<sup>-6</sup> in/in) may be measured.

The so-called "gage factor" (GF) is a calibration factor that relates strain to a change in resistance. The gage factor must be known (it is provided by the gage manufacturer) and the value must be entered into the strain indicator. The course instructor will demonstrate this. The strain indicator is the little "black" box (or blue) that contains circuitry for converting resistance change into a strain reading.



Even large strains will result in only a very small change in the strain gage's resistance; therefore, a Wheatstone bridge is used to measure the resistance change. Wheatstone bridges are electrical circuits that produce a measurable voltage change based on very small changes in resistance. See Figure 1.



Figure 1 - Wheatstone bridge circuit for measuring strain ("quarter bridge")

Notes for Calibrating Cantilever Beam Load Cell

### Materials and Supplies

- Instrumented cantilever beam test apparatus as shown in Figure 2
- P3 strain indicator (not calibrated or certified)
- Set of calibrated or certified masses: 20g, 50g, 100g, 200 g, and 500g
- Clean gloves to be Worn always when handling certified blocks!

NOTE TO SELF: I consider "load" and "weight" to be grams (which is actually a unit of mass).

### Procedure for (unofficial) calibration of cantilever beam load cell:

- 1. Connect the strain gages on the cantilever beam load test apparatus to a strain indicator (if not already done). Enter appropriate gage factors for each gage. Balance the Wheatstone bridges.
- 2. Before collecting data, press down on the end of both cantilever beams to gain a qualitative sense for how much force is required to bend the aluminum and steel beams far enough to just contact the stop. This "real world" experience should help you to evaluate any differences you observe in the data from each beam. When a specific load such as 200g is placed on the aluminum beam, do you expect the strain to be more, less, or equivalent to the strain in the steel beam with the same load?

Calibration involves determining an unknown quantity by comparing with a known quantity. For this calibration, what is unknown is how much strain is produced by loads applied to the ends of the beams.

### 3. Before handling the certified weights, put on clean gloves!

- 4. Place loads shown in Table 1 at the small hole on the end of the steel beam. Measure and record the corresponding strains in Table 1. Keep an eye on the beam's stop do not apply greater loads once contact has been made with the stop.
- 5. Repeat step 4 with the aluminum beam.
- 6. Using the data in Table 1 create a "calibration curve" for each of the two beams. Plot the load (g) as a function of strain ( $\mu\epsilon$ ). This is sort of an odd way to plot the data, because strain is the dependent variable and load is independent; but there is valid reason for plotting it this way. The



slope of the line on the calibration curve is the calibration factor for the load cell and has units of  $g/\mu\epsilon$ . Determine the calibration factor (slope) for both beams (using linear regression curve fit if possible). The calibration factor allows you to determine the applied load for a given strain reading.



Figure 2 - Bending Load Test Apparatus

## Procedure for (unofficially) certifying the three weights as "local standards":

These beams will now be used as load cells; a device used to measure force.

- 1. Using either the steel cantilever beam load cell, determine the "weight" (grams) of the four noncalibrated weights (nominally, 50g, 100g, 200g, 500g) by multiplying the strain reading for each weight by the calibration factor. Record in Table 2.
- 2. Repeat with the aluminum load cell. Record in Table 2.

Different results between the steel and aluminum load cell can be considered part of measurement uncertainty (error).

These weights are now your "local standards". You do not need to wear gloves when handling them, and can "transport" them to the helipads so that you can estimate how precise the helipad measurements are.



# **QUADCOPTER NOTES**

### Definitions

Quadcopter: Something that can be fun to play with in this lab.

**Instrumented helipad**: The transducer (load cell) in this lab is an instrumented beam. The beam deflects an amount proportional to an applied force. The deflection causes a change in resistance in a strain gage, and finally, the strain indicator uses a Wheatstone bridge to convert the change in resistance and gage factor into a strain reading. (It is common for there to be many different parts to a single measurement device).

**Digital scale**: something used to weigh objects. The one in this lab has Velcro to prevent the quadcopter from flying up (used to measure lift produced by the quadcopter).

Notes for Quadcopter Performance Evaluation

Materials:

- An instrumented cantilever beam helipad with P3-strain indicator
- Digital scale (with Velcro) helipad
- The set of four weights you calibrated ("local standards")
- One fully functioning quadcopter
- Safety glasses.

#### Procedures for Performance Characterization

- 1. Put on safety glasses.
- 2. Zero the digital scale. One at a time, use the digital scale helipad to weigh each of the local standards (50g, 100g, 200g, 500g). Enter the data into Table 3 and determine the error between the actual calibrated/certified weight and the measured weight using the digital scale.
- 3. Attach the quadcopter to the Velcro on the digital scale. Record its weight in Table 4. These data provide an estimate of error for the digital scale.
- 4. While the quadcopter is securely attached using the Velcro, gradually increase power until full power is reached. Keep watching the digital read-out change as power is increased. A "zero" reading on the scale indicates that the lifting force balances the quadcopter weight; a negative reading would indicate upward force (lift). If the load indicated on the scale is positive, it indicates that it is not producing enough lift to take off. Record the value for full power "lift" in Table 4 (making sure to include a negative or positive sign as appropriate).
- 5. Zero the strain gage connected to the cantilever beam helipad.
- 6. Place the quadcopter onto the cantilever beam helipad and record the change to the strain reading in Table 5. Using the load cell's calibration factor (as indicated by a sticker on the beam), calculate the lift (multiply strain by the calibration factor). Enter this into Table 5.
- 7. Gently, throttle-up the quadcopter keeping an eye on the strain reading observing how it changes. Once full power is reached, measure and record the strain in Table 5. Using the calibration factor, calculate the lift and enter data into Table 5. If the quadcopter is producing enough lift to "take off" the strain indicated should be negative (upward force). If the strain indicated is positive, it means it indicates that it is not producing enough lift to take off.



### Things I need to investigate (in other words, include in this assignment):

- Free-body diagram of both quadcopter measurement systems.
- What is the percent difference between the weight as measured by the digital scale and the instrumented helipad? How much do they differ? Which one is correct?
- What is the percent difference between the lift as measured by the scale and the instrumented helipad? How much do they differ? Which one is correct?
- Is the difference nearly the same between the lift measurements as compared to the weight measurements or not? Are the differences due to measurement error?

**Comment from conversation with boss**: "Engineers should remember that expert opinion, analysis or even testing do not provide the answer to a design question. They merely provide data to help you make a decision. Decisions require judgment and are based on the available data. What's the importance of engineering judgment when interpreting the results of testing? If you used only the digital scale to measure lift, what might you have concluded about the quadcopters ability to fly? Does there appear to be some value in always questioning data and to view the same problem from different perspectives before judging the validity of your answer?"



 Lab Title:
 \_\_\_\_\_\_Date Conducted:
 \_\_\_\_\_\_Location:

#### Table 1: Strain Value

Nominal Wt.*	Corr. Wt (weight as stated in the certification documents)	6061-T6 Strain	AISI 1015 Strain
(g)	(g)	(με)	(με)
0			
20			
50			
100			
200		( h)	
500			

\*grams are a unit of mass, but are used as "weight" in this laboratory.

Determine calibration factor from the slope of the load vs. strain graph using Table 1 data:

Calibration factors: 6061-T6 load cell = \_\_\_\_\_( $g/\mu\epsilon$ ). AISI 1015 load cell: \_\_\_\_\_( $g/\mu\epsilon$ )

Table 2: Calibration of "Local Standards" (black hook weights)

	AISI	1015	606	1-T6	
Nominal Weight	Measured Strain	Calculated Weight*	Measured Strain	Calculated Weight*	Difference between steel and aluminum load cell (~error)
(8/	(pit)	(8/	(µc)	(8/	18/
50					
100					
200					
500					

\*multiply the measured strain by the appropriate calibration factor to calculate the weight



#### Table 3: Digital Scale Error Estimate

Nominal weight	Calibrated weight	Measured weight	Difference (~error)
(g)	(g)	(g)	<i>(g)</i>
50			
100			
200			
500	•		

Table 4: Digital scale quadcopter results

Description	Measured Load (g)
Powered Off	
Full Power	
Total Thrust	

Table 5: Cantilever beam helipad quadcopter results. Calibration factor noted on the beam:  $g/\mu\varepsilon$ 

Description	Measured Strain $(\mu \varepsilon)$	Measured load (g)
Powered Off		
Full Power		

Photos (as deemed appropriate):

I actively participated in the collection of this data. The information contained here has not been falsified and to the best of my knowledge correctly records the data obtained in the lab.

Print	Name:
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\_\_\_\_\_ Signature:\_\_\_\_\_